

ASME Cyclic Creep Evaluation of Critical Piping Component using CREEP Subroutine and ORNL Test Data

SCIENCE IN THE AGE OF EXPERIENCE - 2016

Presented by:

Brent M. Saba, PE-ME/MT

Saba Metallurgical & Plant Engineering Services, LLC

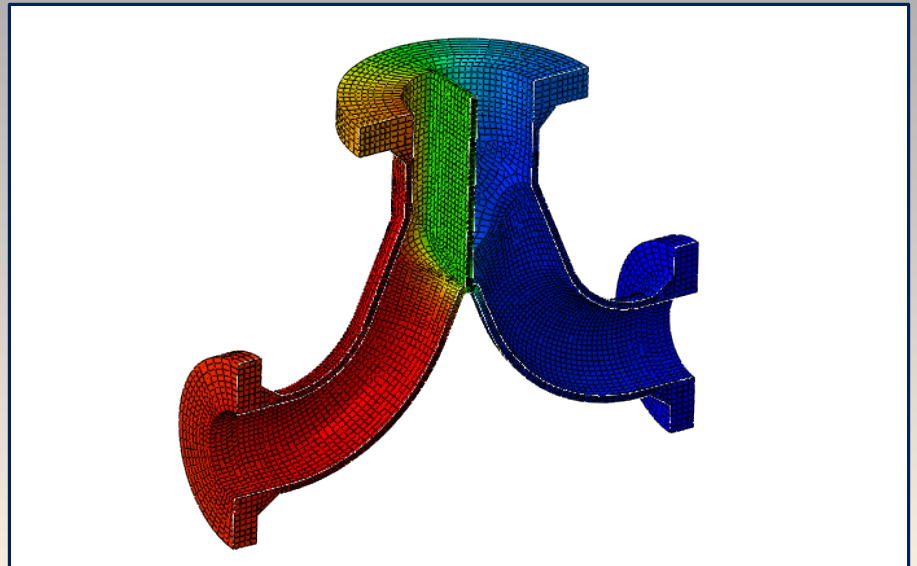
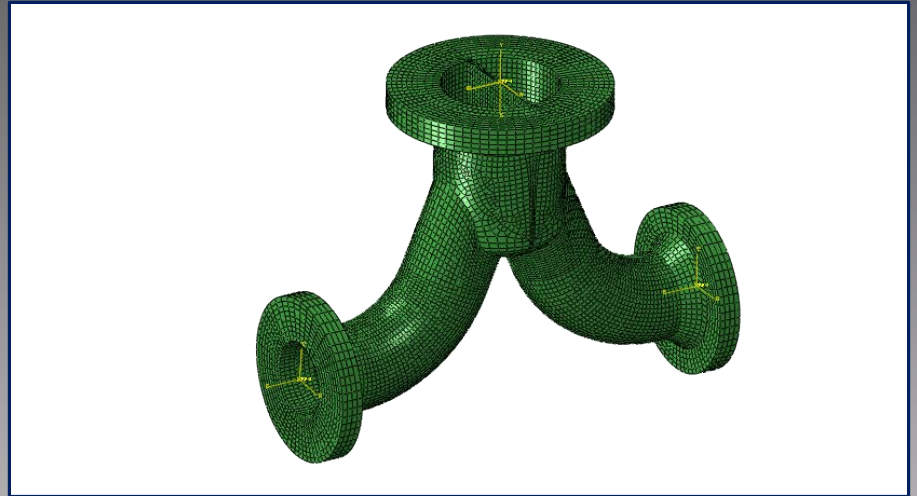
Problem Description

- Cyclic Creep Service – 800-H Material
- ASME Section III-NH (Nuclear Code) Required for Evaluation of Cyclic Creep Conditions
- Primary and Secondary Creep Modeling Necessary – Short Duration Creep Events
- Creep Relaxation Affects Progression of Plasticity
- Use of Abaqus/Standard - Inelastic Analysis

FEA Model Description

- Pipe Wye Component
- Inner Chamber Divider
- Fixed by CaesarII Input
- Displacement Controlled
- Orphan Mesh - Mirrored
- Internal Pressure
- Multi-Stage Cycles

- Flow Alternates
- External Skin Coefficients
- Internal Skin Coefficients
- Ambient Set to 70°F
- Steady Operating Case
- To Flow Case #1
- To Steady Operating Case
- To Flow Case #2



Analysis Procedure

- Run Cycle-By-Cycle Analysis Considering Elastic-Plastic and Creep Effects
- Use CREEP Subroutine to Model Primary and Secondary Stages for 800-H
- Separate Steps for Developing Plasticity and Creep
- Evaluate Cycle-By-Cycle Analysis using ASME Section III-NH (Nuclear) Code

CREEP Subroutine Data

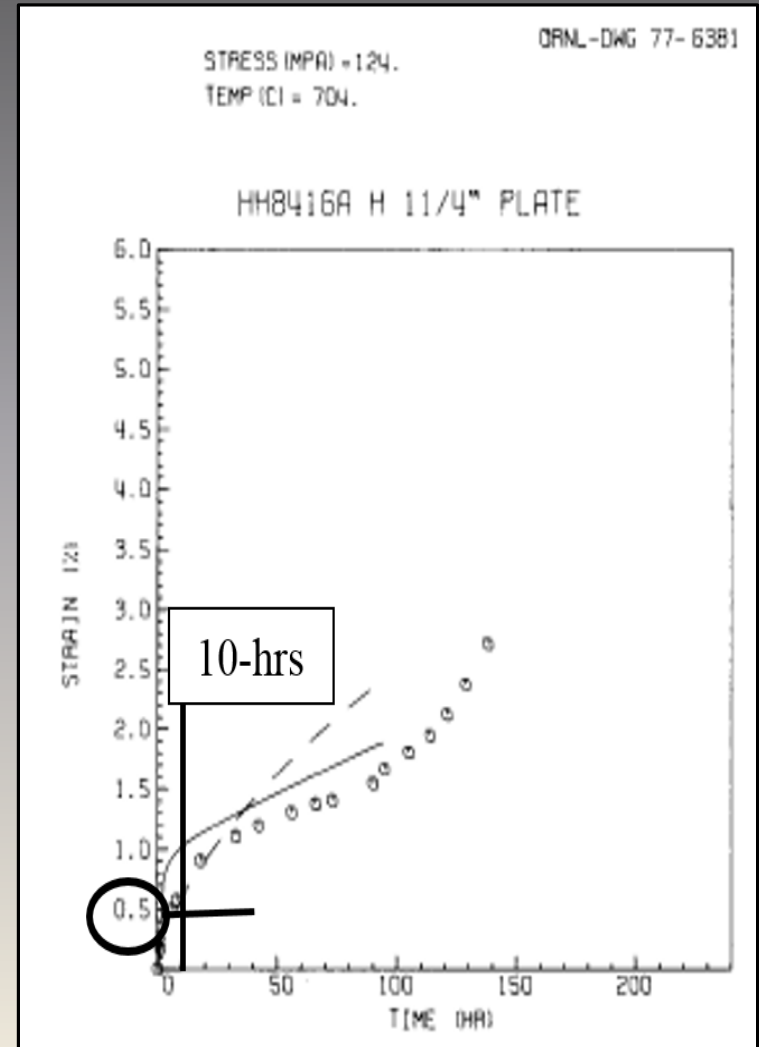
$$\log(\dot{\epsilon}_2) = 24.31 - \frac{40270}{T} + \frac{7040}{T} \cdot \log(\sigma)$$

- Total Creep Strain Rate $f(T, \sigma)$ from ORNL
- $\dot{\epsilon}_2$ = strain rate (primary and secondary)
- T = temperature (°C)
- σ = stress (MPa)

Survey of Available Creep and Tensile Data for Alloy 800H by M.K. Booker, V.B. Baylor, and B.L.P. Booker, a production of Oak Ridge National Laboratory (ORNL), 1978

Over 100+ raw data plots of strain versus time, at defined temperatures and stresses, using various plate thicknesses.

Creep initiation considered instantaneous



CREEP Subroutine - 1

```
800H_creep_english.for  X
(Global Scope)
SUBROUTINE CREEP(DECRA, DESWA, STATEV, SERD, EC, ESW, P, QTILD,
1 TEMP, DTEMP, PREDEF, DPRED, TIME, DTIME, CMNAME, LEXIMP, LEND,
2 COORDS, NSTATV, NOEL, NPT, LAYER, KSPT, KSTEP, KINC)
C
C   INCLUDE 'ABA_PARAM.INC'
C
C   CHARACTER*80 CMNAME
C
C   DIMENSION DECRA(5), DESWA(5), STATEV(*), PREDEF(*), DPRED(*),
1   TIME(3), COORDS(*), EC(2), ESW(2)
C
C   DEFINE CONSTANTS
C   UNITS ARE ENGLISH -- PSI, Farenhite, Hours
C
C   A=24.31D0
C   B=40240.0D0
C   C=7040.0D0
C   TEMP_CUTOFF=1000.0D0
C   T=TEMP
```

$$\log(\dot{\epsilon}_2) = 24.31 - \frac{40270}{T} + \frac{7040}{T} \cdot \log(\sigma)$$

ORNL Equation NOT Valid Below 1000°F

A = 24.31
B = 40.270
C = 7040

CREEP Subroutine - 2

```
C    FACTOR1 AND FACTOR2 NEED TO BE COMPUTED FOR MODIFYING  
C    QTILD AND T  
      FACTOR1=6.8947D-3  
C    T CHANGED FROM FARENHITE TO CELSIUS  
      T1=(T-32.0)*5.0D0/9.0D0
```

FACTOR1 – Conversion from MPa to psi

T1 – Conversion from °C to °F

QTILD – Equivalent Deviatoric Mises Stress, σ

```
C TO ADDRESS MINIMAL CREEP OCCURRING BETWEEN TEMP 900-1100  
      DECRA(1) = 1.D-10  
      DECRA(5) = 1.D-30  
C    WRITE(6,*) 'DECRA(5), T', DECRA(5), T
```

DECRA(1) – Equivalent (Uniaxial) Deviatoric Creep Strain Increment = $\text{EXP}(F) \cdot \text{DTIME}$

DECRA(5) – Partial Derivative of Strain Rate with Respect to QTILD

DTIME – Time Increment (hrs)

CREEP Subroutine - 3

```
IF(T.GE.TEMP_CUTOFF) THEN
C   QTILD, NEEDS TO BE MODIFIED AS SHOWN BELOW
   IF(QTILD.LE.0.0D0) QTILD=1.D-2
   QTILD1 = QTILD*FACTOR1
C   WRITE(6,*) 'QTILD', QTILD1, QTILD
   F=A-B/T1+(C/T1)*LOG(QTILD1)
C   WRITE(6,*) 'VALUE', F
C   STOP
C   PRINT*, F
DECRA(1) = EXP(F)*DTIME
```

QTILD can not be 0 or negative

Virtually no creep in compression

$F = \text{Log}(\dot{\epsilon}_2)$ - Taken straight from ORNL Eqn

DECRA(1) - Equivalent (Uniaxial) Deviatoric
Creep Strain Increment = $\text{EXP}(F)*\text{DTIME}$

DECRA(1) - Strain over a differential length
of time

CREEP Subroutine - 4

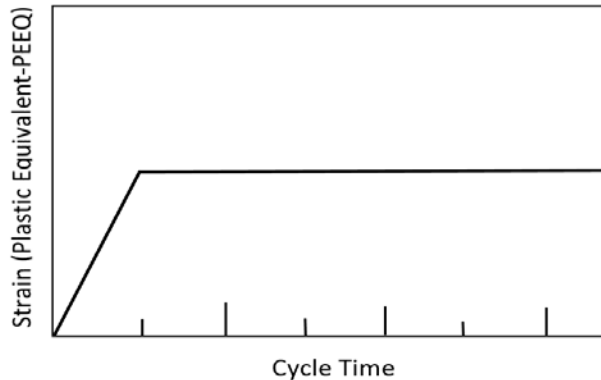
```
      IF(LEXIMP.EQ.1) THEN
C      DECRA(5) = 1.D-10
C      IF(QTILD1.GE.1.0D-5) THEN
        DECRA(5) = (DTIME*EXP(F)*C/(T1*QTILD1))*FACTOR1
C      WRITE(6,*) 'DECRA(5) & 1', DECRA(5), DECRA(1), QTILD1
C      END IF
      END IF
      END IF
C
      RETURN
      END
```

Implicit Condition Statement

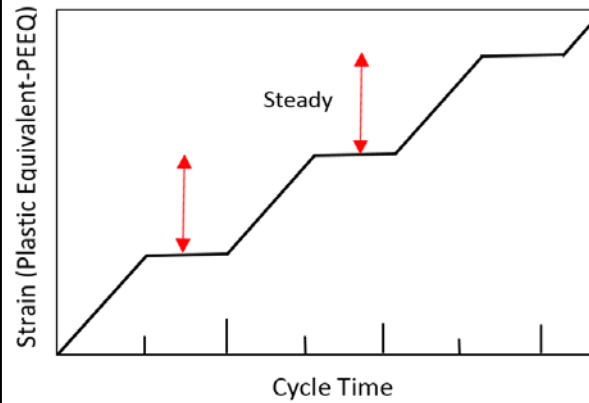
DECRA(5) - Partial Derivative of Strain Rate
with Respect to QTILD

Shakedown and Ratcheting

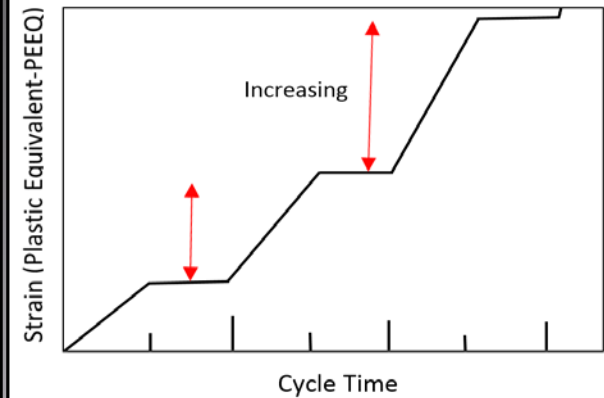
Elastic Shakedown



Plastic Shakedown



Ratcheting



Elastic Shakedown

- One Time Plastic Event
- Twice Yield Limit

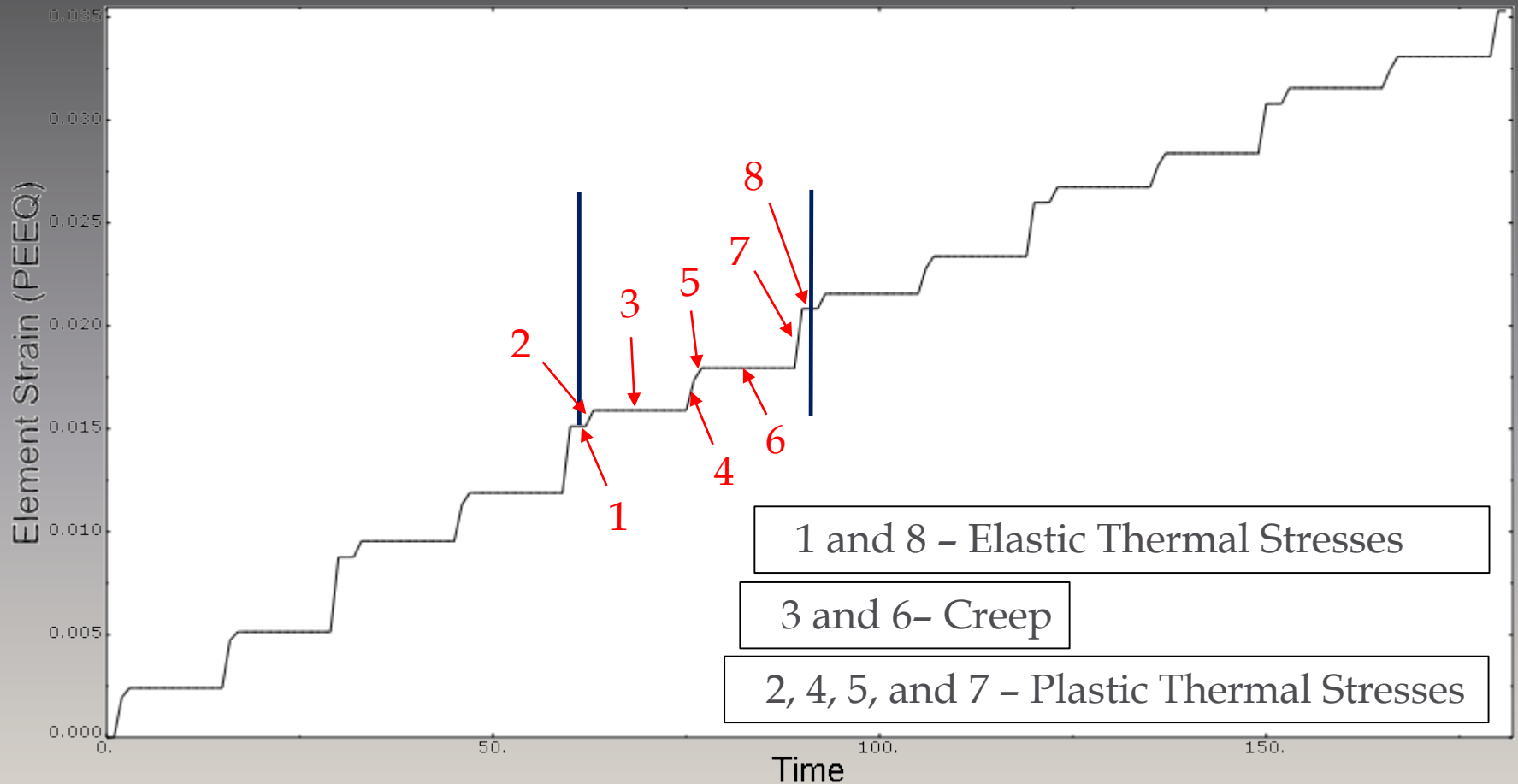
Plastic Shakedown

- Reoccurring constant strain amplitude
- Displacement Controlled Loading

Ratcheting

- Increasing Strain Per Cycle
- Load-Controlled (Requires Pressure)

Results – Plastic Accumulation



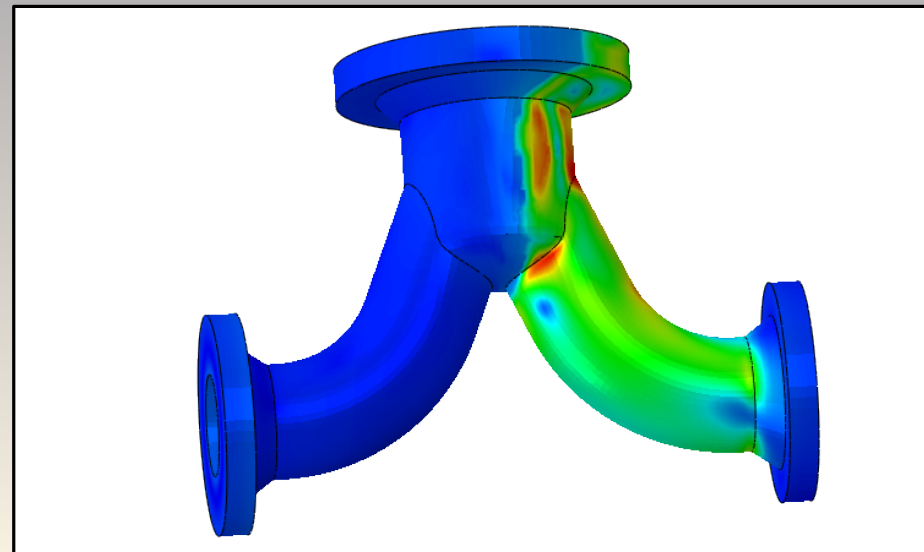
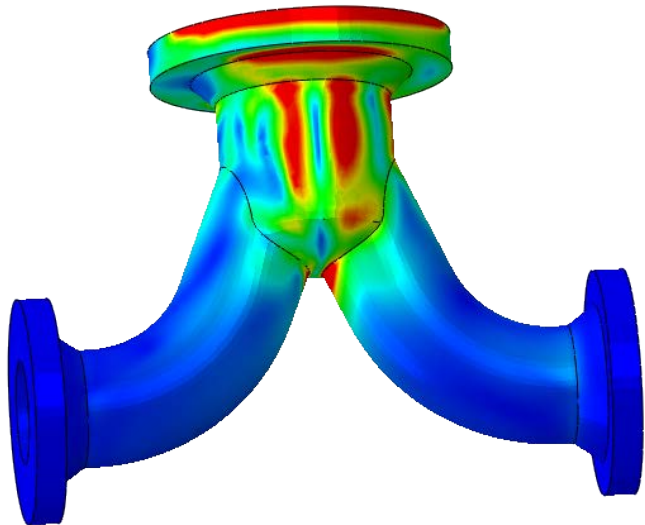
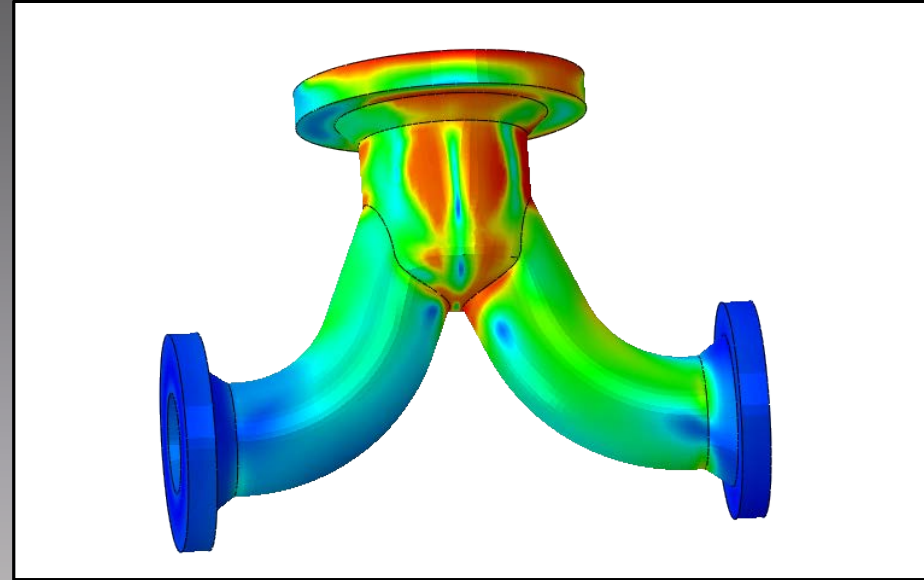
- 1) **1)** Ambient-Operating (No Strain after 1st), **2)** Operating-Leg1HighTemp (Strain Slight Ramp-Up), **3)** Creep-Leg1 (No Strain) **4)** Leg1HighTemp-Operating (Strain Ramp), **5)** Operating-Leg2HighTemp (Strain Slight Ramp-Up), **6)** Creep-Leg2 (No Strain), **7)** Leg2HighTemp-Operating (Strain Ramp), ReturntoAmbient (NoStrain)

Creep Relaxation

One leg subjected to high temperature and resultant creep relaxation. Then repeats for opposite leg

Some creep relaxation of opposite leg, but maximum stress on that leg remains.

Back to operating causes additional strain accumulation.



Equivalent Strain Range ASME III-NH-T-1413

$$\Delta\epsilon_{equiv,i} = \frac{\sqrt{2}}{2(1 + \nu^*)} \left[(\Delta\epsilon_{xi} - \Delta\epsilon_{yi})^2 + (\Delta\epsilon_{yi} - \Delta\epsilon_{zi})^2 + (\Delta\epsilon_{zi} - \Delta\epsilon_{xi})^2 + \frac{3}{2} (\Delta\gamma_{xyi}^2 + \Delta\gamma_{yzi}^2 + \Delta\gamma_{zxi}^2) \right]^{\frac{1}{2}}$$

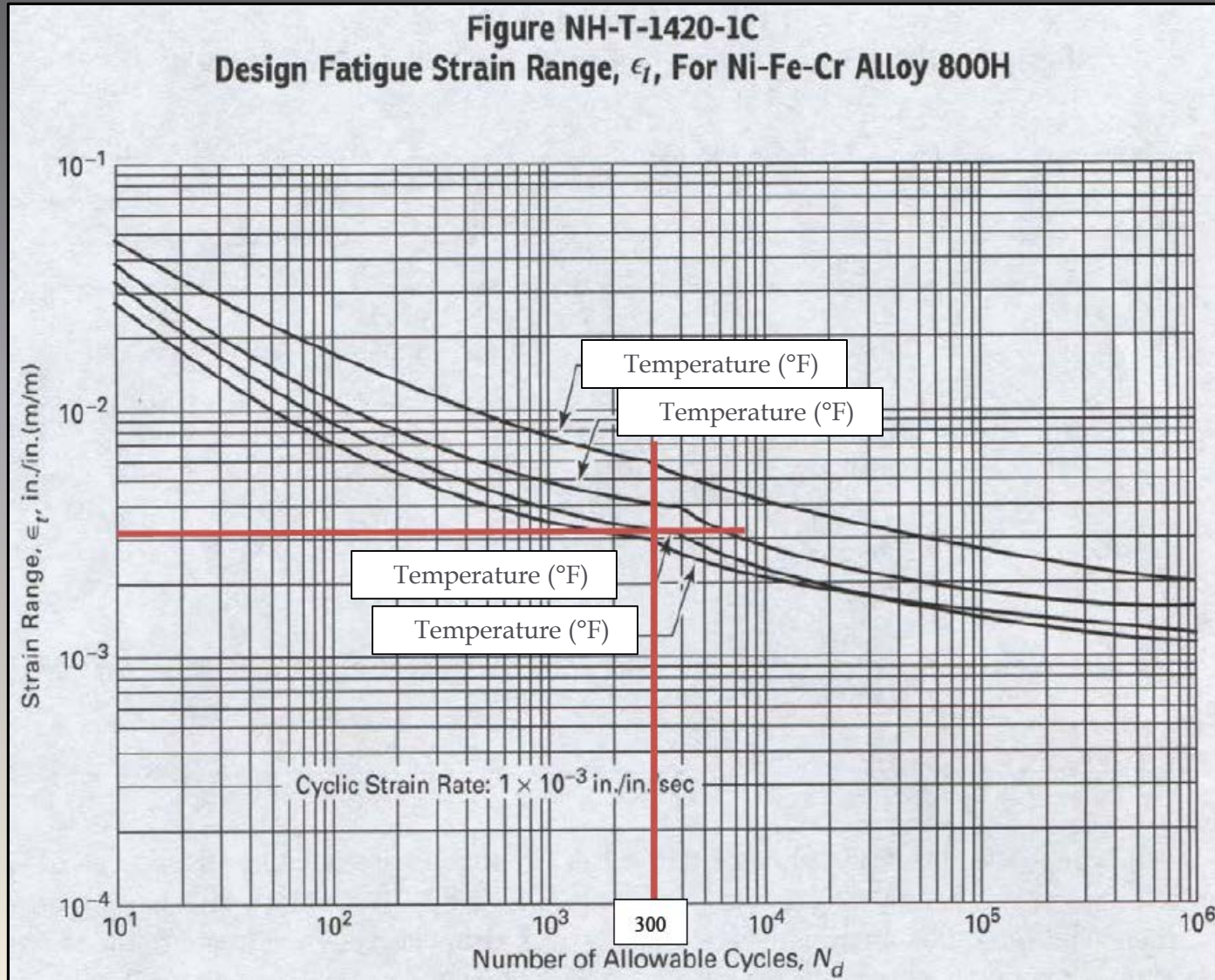
1. Determine, at nodes of interest, all six strain components ($\epsilon_{x,y,z}$ and $\gamma_{xy,yz,zx}$).
2. Determine extreme (maximum or minimum); designate as "o".
3. Calculate each change in strain at each node at each internal cycle inflection.

$$\Delta\epsilon_{xi} = \epsilon_{xi} - \epsilon_{xo}$$

$$\Delta\gamma_{xyi} = \gamma_{xyi} - \gamma_{xyo}$$

4. Calculate the equivalent strain range for each internal cycle point (Top Eqn).
5. Define $\Delta\epsilon_{max}$ as the maximum value from Step 4.

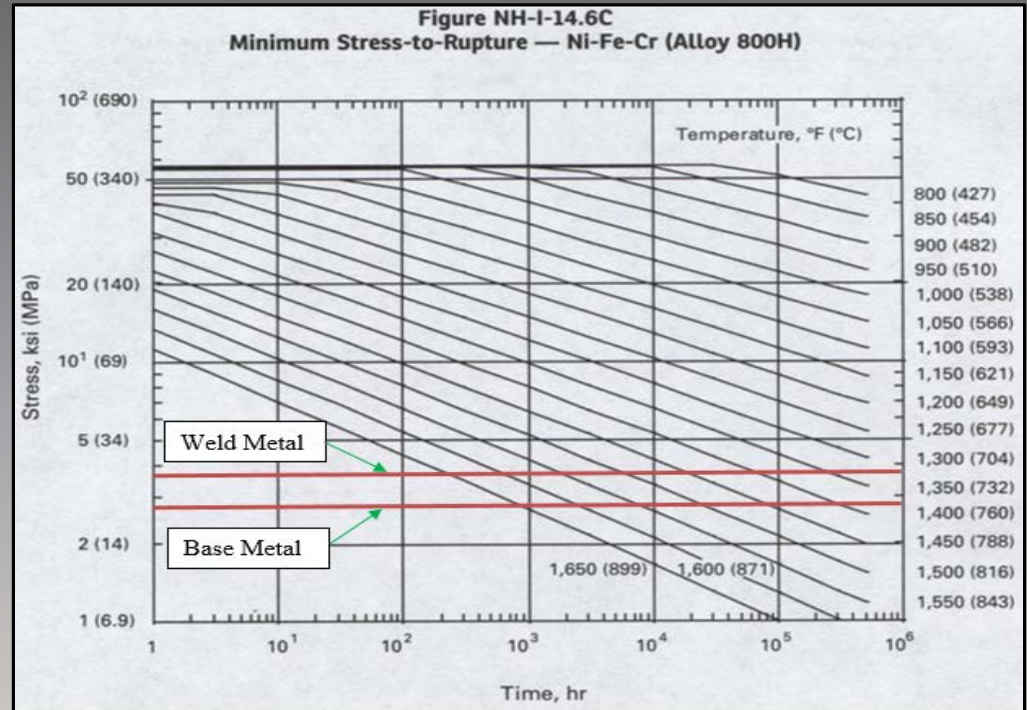
ASME III-NH Evaluation Curve



Minimum Stress-to-Rupture ASME III-NH Figure NH-I-14.6C

Weld metal (Inco-82, SFA-5.14 ERNiCr-3) stress rupture factor is a divisor of 0.8 per Table NH-I-14.10C-2.

Creep is insignificant based on calculated stress values.



Averaged maximum principal stress over the last cycle used to calculate von Mises Stress Intensity, divided by K' (0.67 for Alloy 800-H per Table NH-T-1411-1). For inelastic analysis with Alloy 800-H, multiplier factor to SI is 1.0.

This stress equivalent/ K' value is used in Figure NH-I-14.6C to determine allowable time duration (T_d).

Creep-Fatigue Analysis

ASME III-NH-T-1411

$$\sum_{j=1}^p \left(\frac{n}{N_d} \right)_j + \sum_{k=1}^q \left(\frac{\Delta t}{T_d} \right)_k \leq D$$

Per NH-T-1715 (Creep-Fatigue Reduction Factors), the N_d value in vicinity of a weld shall be one-half the value permitted for the parent material.

n = number of applied cycles

N_d = number of design allowable cycles (Slide 14)

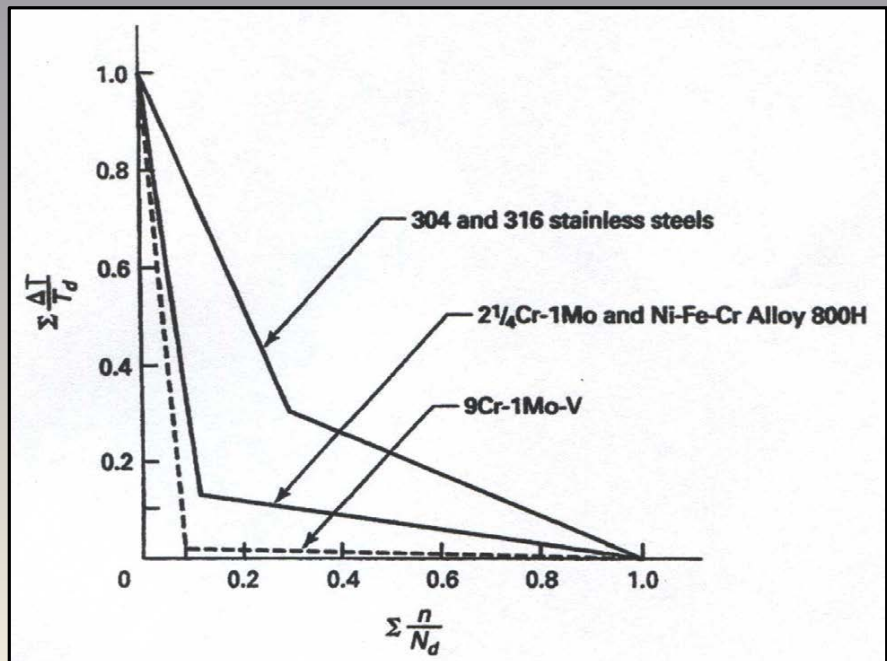
Δt = duration of time at creep

T_d = allowable duration at creep temperature

D = total creep-fatigue damage, inside the boundaries of Figure NH-T-1420-2 (to right)

Creep term is 0.

Solve for n .



Conclusions

- CREEP Subroutine necessary to properly capture short term primary creep effects

- With creep relaxation, creep stress was low enough to make creep damage insignificant

- Creep and creep relaxation, however, did affect the plastic strain range

- ASME III-NH (High Temperature Nuclear Code) is used to evaluate an allowable number of operating cycles

Acknowledgements

Mr. Shashwat Sinha, Senior Technical Specialist with Simulia, provided direct training and was an integral part of the development of the CREEP Subroutine Fortran program. His efforts are greatly appreciated.

A special shout-out to our Abaqus Unified FEA reseller, Mr. Gabe Dambaugh of FEA Services, for his continual excellent servicing commitment.

Thank You

Questions or Comments?